

## Center for Advanced **Non-Ferrous Structural Alloys** An Industry/University Cooperative Research Center

### **Project # 35:On the Influence of Microstructural** Features of Linear Friction Welding and Electron **Beam Additive Manufacturing Ti-6AI-4V on Tensile and Fatigue Mechanical Properties**

Spring 2019 Semi-Annual Meeting Iowa State University, Ames, IA April 3-5, 2019

Student: Michael Mendoza (ISU) Faculty: Peter Collins ISU Industrial Mentors: Honeywell





Project 35: On the Influence of Microstructural Features of Linear Friction Welding and Electron Beam Additive Manufacturing Ti-6AI-4V on Tensile and Fatigue Mechanical Properties



<ul> <li>Student: Michael Mendoza (ISU)</li> <li>Advisor(s): Peter Collins (ISU)</li> </ul>	Project Duration PhD: January 2017 to July 2019
<ul> <li><u>Problem:</u> Linear Friction Welding (LFW) offers cost reduction for aircraft structural components production. However, the information about its microstructure and mechanical properties is still limited.</li> <li><u>Objective:</u> Characterize local microstructures (LFW) and their relationship with mechanical properties</li> <li><u>Benefit:</u> The understanding of microstructure-properties relationship of LFW will improve manufacturing efficiency of aircraft components.</li> </ul>	<ul> <li>Recent Progress</li> <li>Larger EBSD characterization of The Welded Zone (WZ), WZ in-plane, Thermomechanical Affected Zone (TMAZ) and Parent Material (PM)</li> <li>Additional tensile tests and stress/strain curves on dogbone samples for the individual zones of LFW</li> <li>Preliminary ultrasonic fatigue test on Branson device for purchase and modifications</li> <li>PED-TEM data for dislocation density calculations</li> </ul>

Metrics			
Description	% Complete	Status	
1. Literature review	95%	•	
2. Microstructure and tensile properties of dogbone samples within the individual three LFW zones	100%	•	
3. Conventional fatigue analysis (four-point bending test) of local microstructures EBAM-Ti-6AI-4V	80%	•	
4. Simulation and design (Comsol) of ultrasonic fatigue on local microstructures of EBAM-Ti-6AI-4V		•	
5. Ultrasonic fatigue test design and modifications	20%	•	

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### **Industrial Relevance**



- The study of Ti-6AI-4V under different manufacturing processes is attracting more interest from industry because of cost reduction and potential improvements in mechanical properties.
- The main advantage of LFW resides in the fact that for aircraft structural components oversized ingots are machined to get the final component, so a large amount of material is wasted. LFW allows the use of not oversized ingots for welding them together to form the component with less use of initial material.

### Linear Friction Welding (LFW)Ti-6AI-4V



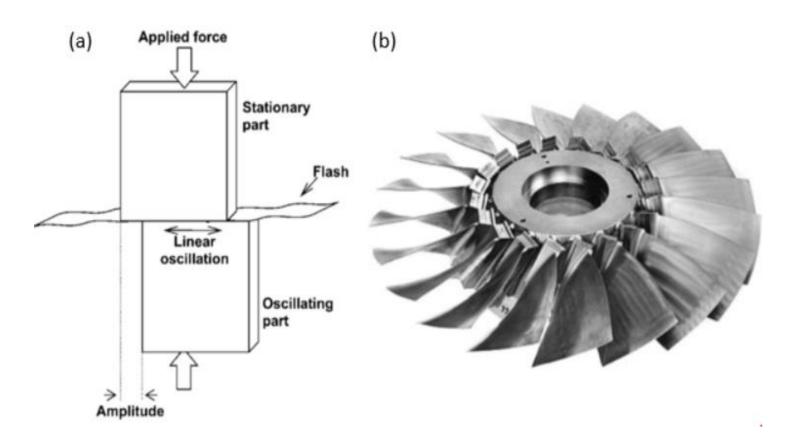


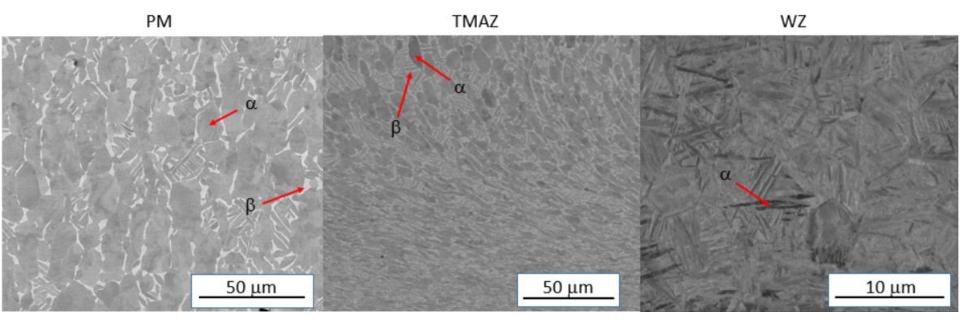
Figure 1.(a) Diagram of Linear Friction Welding process, (b) Integrated blisk (disc and blades)

Bhamji, I., Preuss, M., Threadgill, P. L., & Addison, A. C. (2011). Solid state joining of metals by linear friction welding: a literature review. *Materials science and technology*, 27(1), 2-12.

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# Linear Friction Welding (LFW) Zones





Backscatter electron micrographs on each zone

- PM parent material with a bi-modal microstructure (i.e. primary  $\alpha_p$  grains surrounded by  $\alpha$  lamellar microstructure of  $\alpha$  laths in a  $\beta$  matrix).
- TMAZ -Thermomechanical affected zone with a distorted bi-modal microstructure.
- WZ Welded zone with a refined martensitic  $\alpha$ ' (needle like) laths in a  $\beta$  matrix.

### Linear Friction Welding (LFW) Zones

### **IPF Z Color**



1210

0110

0001

PM TMAZ WZ

<u>10 μm</u>

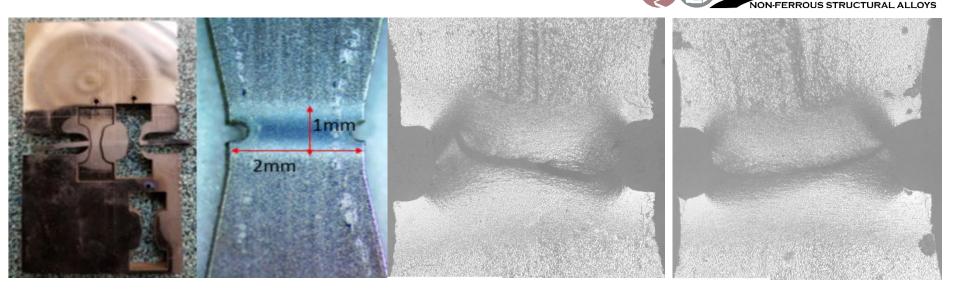
#### Backscatter electron micrograph

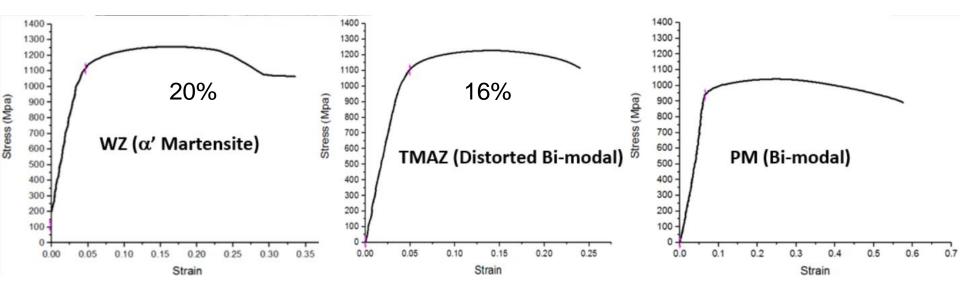
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100µm

### Current Progress on Tensile Properties of LFW-Ti-6AI-4V





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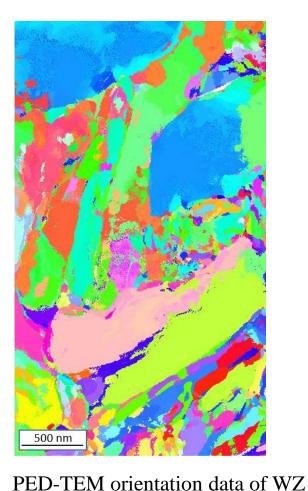
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### **Dislocation Density Analysis**





PED-TEM orientation data of TMAZ for dislocation density analysis

Orientation information is the input file for the MATLAB code for dislocation density calculation

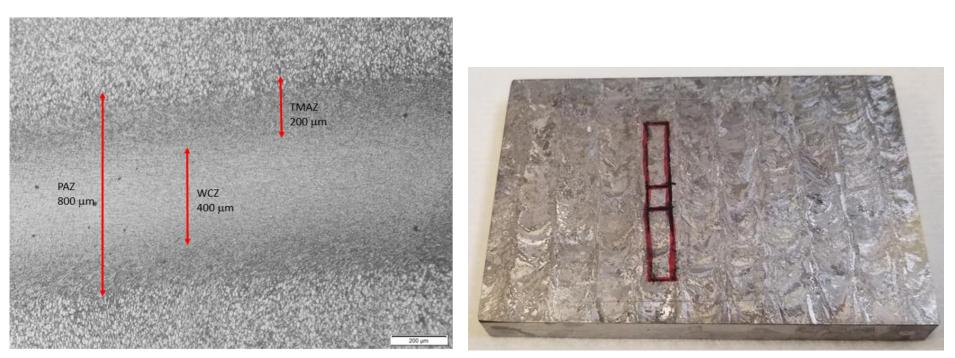
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for dislocation density analysis

# **Fatigue Approach**



Tensile properties are important, but fatigue analysis is also necessary due to the presence of cyclic stresses on Ti-6AI-4V aircraft components.

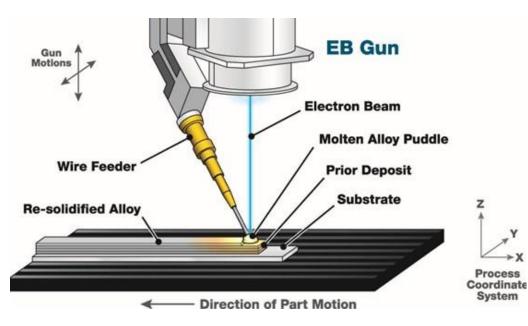


EBAM-Ti-6Al-4V

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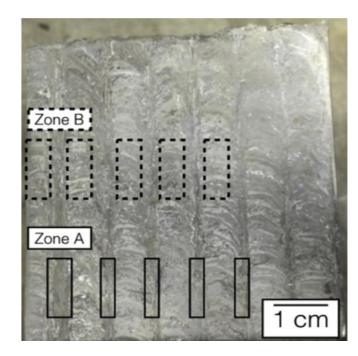
### **Electron Beam Additive Manufacturing** (EBAM)





http://www.sciaky.com/additive-manufacturing/wire-am-vs-powder-am

- Electron beam as a heat source
- Ti-6AI-4V wire as a feedstock
- Vacuum chamber that also protects the alloy



• Two distinct zones can be recognized on this picture, zone A comprises vertically elongated prior  $\beta$  grains with very little variation in  $\alpha$  lath thickness and zone B with a pronounce variation in  $\alpha$  lath thickness and a more scattered orientation.

Hayes, B. J., et al. (2017). "Predicting tensile properties of Ti-6Al-4V produced via directed energy deposition." Acta Materialia 133: 120-133.

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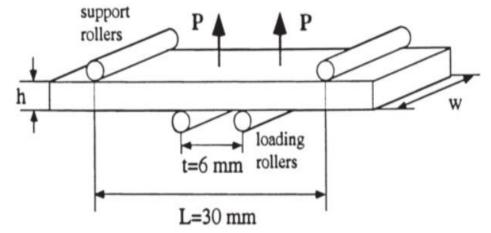
### **Conventional Fatigue Analysis on EBAM-Ti-6AI-4V**



Four-point bending test is selected as a convenient method for fatigue studies due to several reasons:

- It produces a uniform maximum stress on the surface.
- Easy sample mounting and dismounting as no special gripping is required.
- It is also suitable to evaluate specific microstructures from small samples.

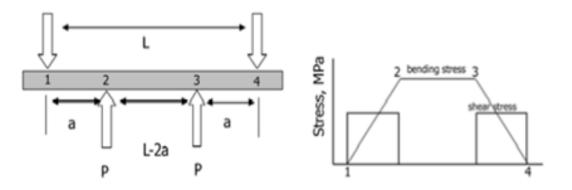
20 specimens total (Length: 40 mm, width: 5 mm, Thickness: 4.5 mm) 10 for zone **A** and 10 for zone **B** were sectioned via EDM with the suitable dimensions for capturing the interested microstructure and construct the respective *S-N* curve and fracture analysis.



T. Zhai, Y. Xu, J. Martin, A. Wilkinson, G. Briggs, A self-aligning four-point bend testing rig and sample geometry effect in four-point bend fatigue, International Journal of Fatigue 21(9) (1999) 889-894.



# **Four-Point Bending Test**



Optimum testing geometry for uniform stress distribution consistent with the value calculated by the beam theory.

t = load span L = support span h = thickness

> t/h = (5.74 mm)/(4.5 mm) = 1.27L/t = (23 mm)/(5.74 mm) = 4.0





Pilchak, A. L. (2009). The effect of friction stir processing on the microstructure, mechanical properties and fracture behavior of investment cast Ti-6AI-4V, The Ohio State University.

T. Zhai, Y. Xu, J. Martin, A. Wilkinson, G. Briggs, A self-aligning four-point bend testing rig and sample geometry effect in four-point bend fatigue, International Journal of Fatigue 21(9) (1999) 889-894.

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# Comsol Model of Ultrasonic Fatigue on EBAM-Ti-6AI-4V

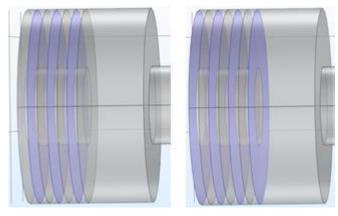


Catenoidal ONE-HALF WAVELENGTH\* HORN CONVERTER BOOSTER **Piezoceramics** 5.57 Thickness = 0.2 in INCHES 7.16 182 141 5±0.5 127±12.7 MM 0 < Ó 3.50 00 401 CU 0.8 in RECOMMENDED CLAMPING AREA **OVERALL HORN LENGTH** CAN VARY BEYOND THESE BOOSTER FRONT END TYPICAL DIMENSIONS DIA. WILL VARY WITH DEPENDING ON THE APPLIC TION AMPLITUDE UNING A \*\*DIMENSIONS VARY WITH 7 50 5 mm 0 Ø 500 µm 100 Horn (Catenoidal 3/8") 50 -200 Booster Converter 0 -100 133 mm mm 139.1 mm 101.5 mm 0 -50 mm

20kHz Converter/Booster/Horn, Typical Dimensions

Branson Ultrasonics Corporation.

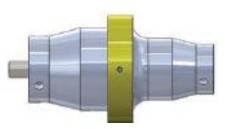
### **Resonance system**



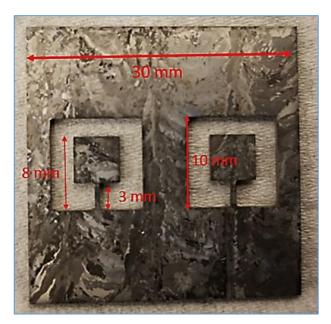
### Electric potential (V) Ground

6 piezoceramic discs, stacked mechanically in series and electrically in parallel.



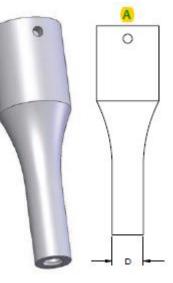


Gold 1:1.5 101-149-057 Booster for 20kHz

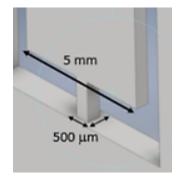




#### Catenoidal Horn



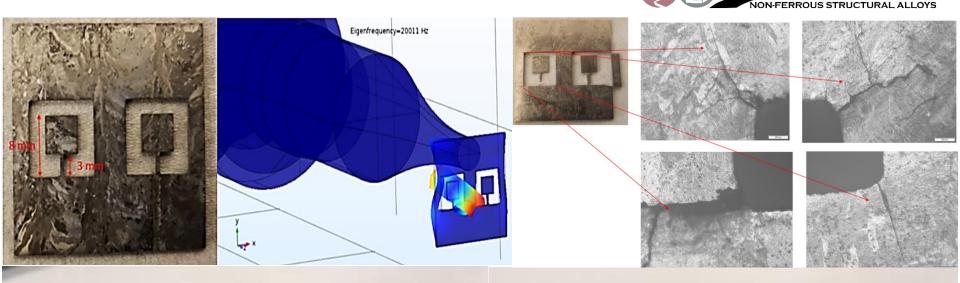
### 608-001-021

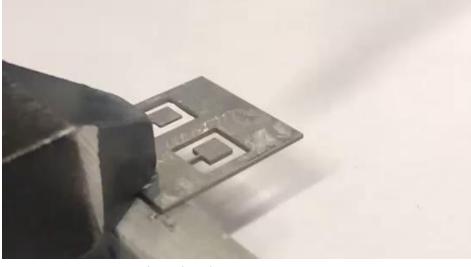


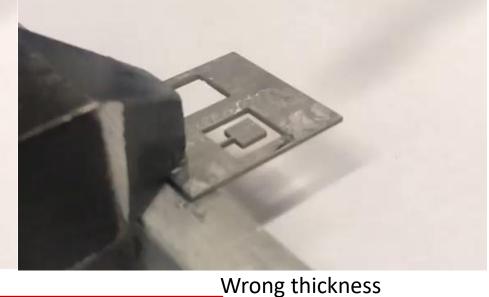
Sample dimensions

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### Ultrasonic Fatigue Prediction and Test on CANFSA EBAM-Ti-6AI-4V







#### **Right thickness** SEMI-ANNUAL MEETING – 4/3/2019

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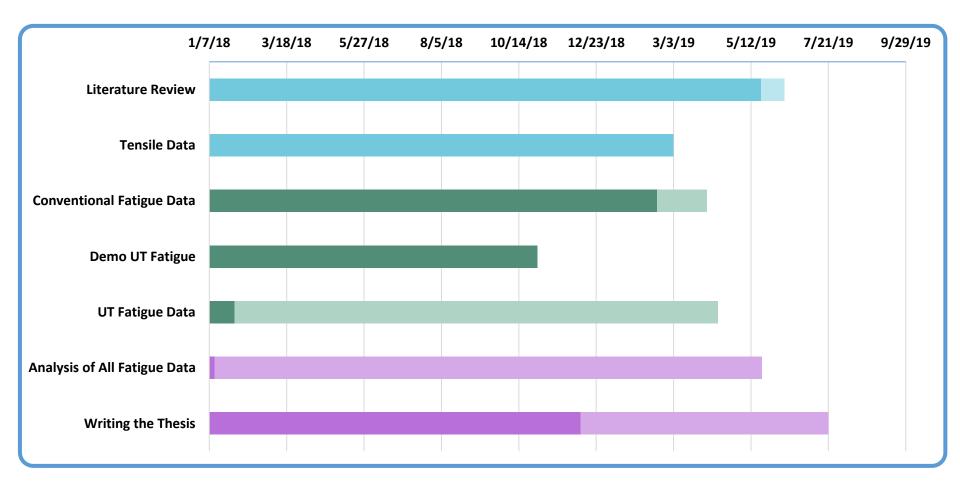
# **Summary of Pending Work**



- TEM-PED data and MATLAB code for dislocation density calculations
- S-N curve construction and analysis from conventional fatigue test
- Branson Ultrasonic equipment modifications
- S-N curve construction and analysis from ultrasonic fatigue test
- Analysis and comparison of the fatigue tests

### **Progress Gantt Chart**





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### Center for Advanced **Non-Ferrous Structural Alloys** An Industry/University Cooperative Research Center

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# Thank you very much!

## Michael Mendoza

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# Center for Advanced **Non-Ferrous Structural Alloys** An Industry/University Cooperative Research Center

**Project 35 - On the Influence of Microstructural Features of Linear** Friction Welding and Electron Beam Additive Manufacturing Ti-6AI-4V on Tensile and Fatigue Mechanical Properties

Student: Michael Mendoza

Faculty: Peter Collins

Industrial Partners: Honeywell

Project Duration: January. 2017 – July 2019

#### **Achievement**

Characterize local microstructures of Linear Friction Welding (LFW) and their relationship with mechanical properties.

#### Significance and Impact

New welding methods as LFW offers cost reduction for aircraft structural components production. Understanding the microstructure-properties relationship in the process is a key factor to its implementation.

#### **Research Details**

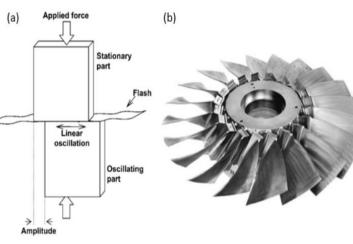
Microstructure characterization of individual LFW-Ti-6AI-4V zones to evaluate tensile properties and exploration of fatigue analysis on larger local microstructures as EBAM-Ti-6AI-4V for future applicability on LEW.

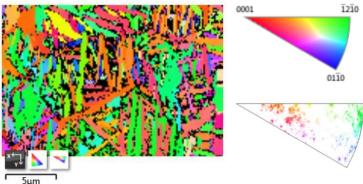












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#### **Significance and Impact**

Student: Michael Mendoza

New welding methods as LFW offers cost reduction for aircraft structural components production. Understanding the microstructure-properties relationship in the process is a key factor to its implementation.

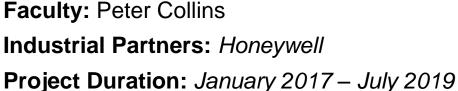
#### **Research Details**

Achievement

properties

Microstructure characterization of individual LFW-Ti-6AI-4V zones to evaluate tensile properties and exploration of fatigue analysis on larger local microstructures as EBAM-Ti-6AI-4V for future applicability on LFW.

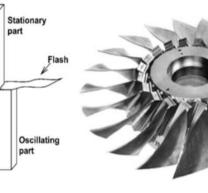
# Project 35 - Characterization of Microstructures and Mechanical Properties in LFW Ti-6AI-44 CANFSA



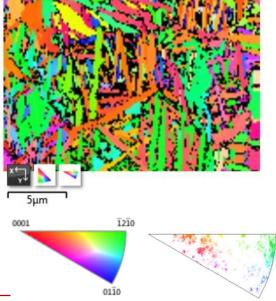
Linear Characterize local microstructures of Linear Friction oscillation Welding (LFW) and their relationship with mechanic nart

(a)

Applied force



(b)



### Project 35 - Characterization of Microstructures and Mechanical Properties in LFW Ti-6AI-4V

#### Student: Michael Mendoza

Faculty: Peter Collins

Industrial Partners: Honeywell

Project Duration: January. 2017 – July 2019

#### Program Goal

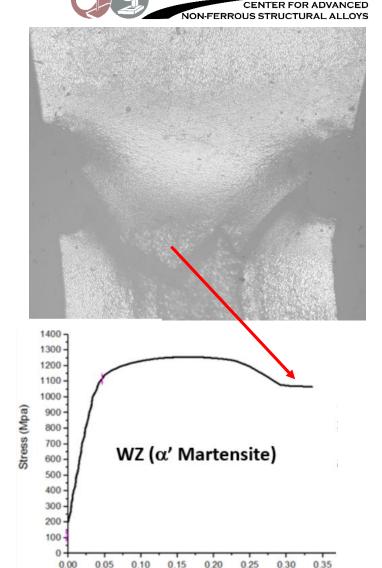
 Characterize the microstructure and mechanical properties of Linear Friction Welding (LFW)

#### **Approach**

• Evaluate tensile properties on LFW-Ti-6AI-4V and explore fatigue analysis on larger microstructures as EBAM-Ti-6AI-4V for future applicability on LFW.

#### **Benefits**

 The understanding of microstructure-properties relationship of LFW will improve manufacturing efficiency of aircraft components.



Strain

CANESA